Shortening the Insertion Time for Materials Technologies— the 21st Century Metals Challenge



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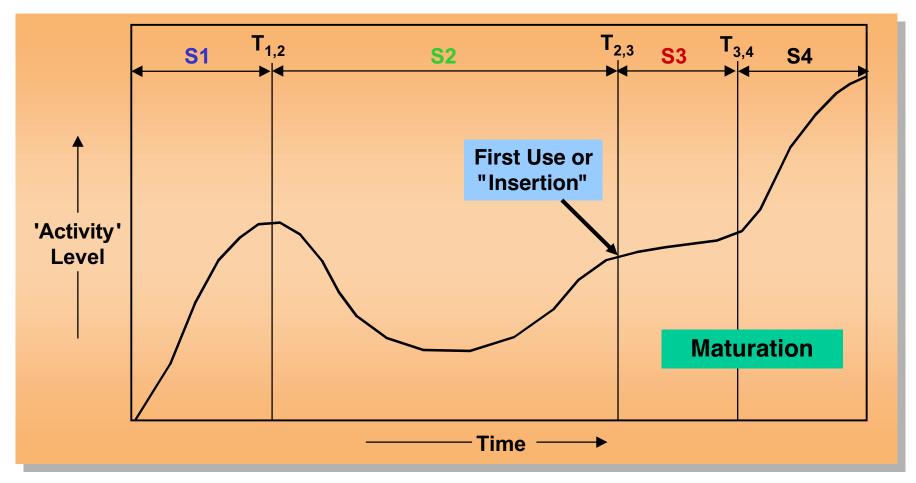
Leo Christodoulou, Steve Wax





Observation: Historical Aerospace Material Life Cycle





Stages

S1= Revolutionary

S2 = Emerging

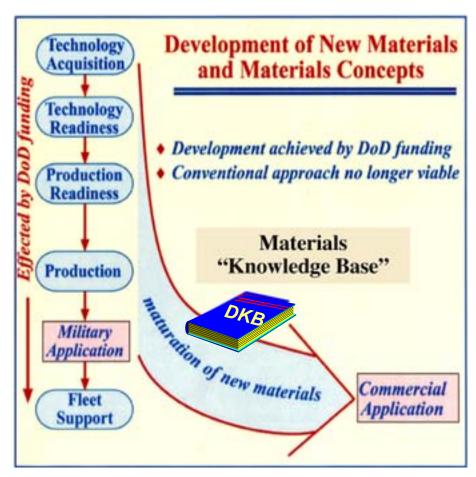
S3 = Specialty Material

S4 = Commodity Material

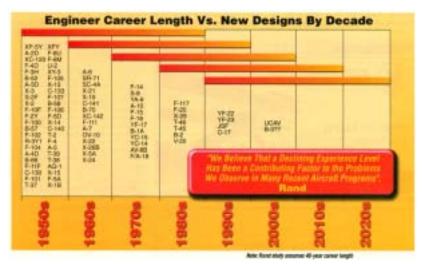


Aerospace Structural Materials Development: How It Happened





Adapted from Fraser, 1998; Wax, 1999



- DoD materials transition opportunities (systems) have drastically reduced
- Material development time far exceeds the modern short product cycle
 - iterative, empirical development of "Knowledge Base" is lengthy, data intensive, and expensive

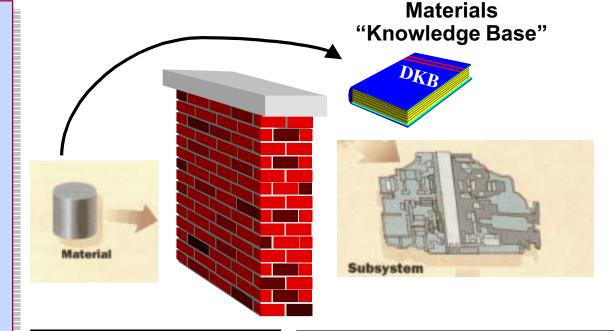


The Disconnect!



Major disconnect between materials development & components/systems engineering design

- Known alloy to reliable part ~36 months
- Steels for navy landing gear 15+ yrs
- Lightweight composites for army vehicles 15+ yrs
- Gamma titanium aluminides ~30yrs and counting
- Ceramics for engines -30+++? yrs
- Evolutionary alloy changes (ship steels, superalloys, etc) ~7-10 years



Materials

Development

- Highly Empirical
- Testing Independent of Use
- Existing Models Unlinked



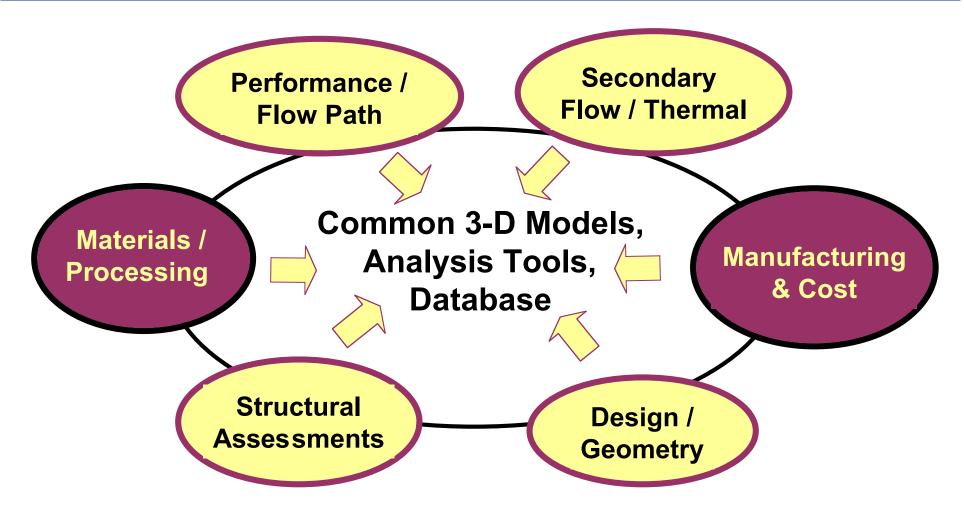
Engineering Design

- Materials Input from "Knowledge Base" of Data (Data Sheets, Graphs, Heuristics, Experience, etc.)
- System/Sub-System Design is Heavily Computational and Rapid
- Well Established Testing Protocols



Integrating Materials & Processes with Engine Design





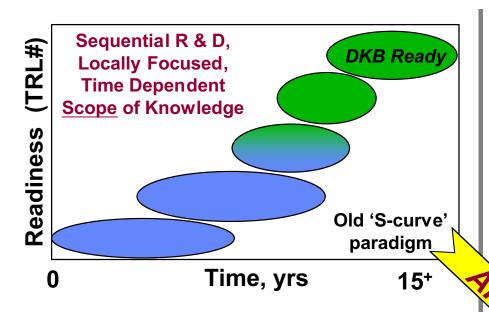
Design "development cycle": <3 yrs

Materials & Process "cycle": 7-20 yrs



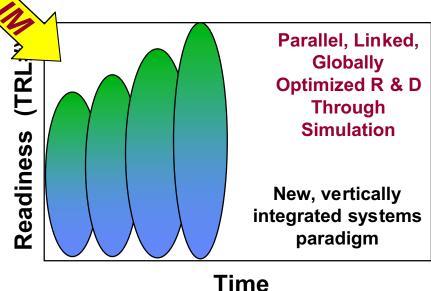
AIM Paradigm for Materials R & D





- Building "Designer Knowledge Base" begins at outset
- Optimization based on <u>design</u>
 IPT need
- Time & effort refines <u>quality</u> of knowledge base, <u>not its scope</u>

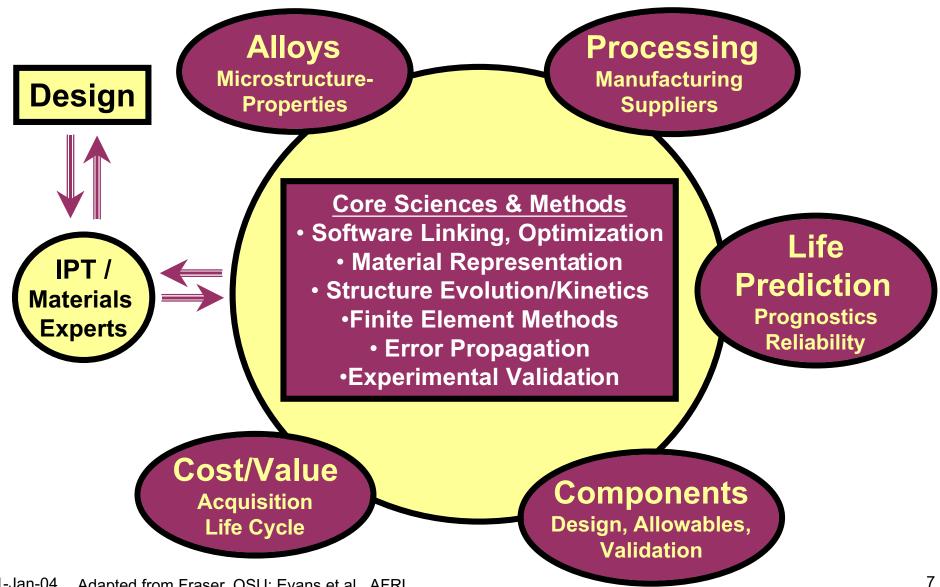
- Sequential M & P
- Optimized from heuristics
- "Designer Knowledge Base"
 NOT Ready Until Final Stages





Major Components of Designer Knowledge Base







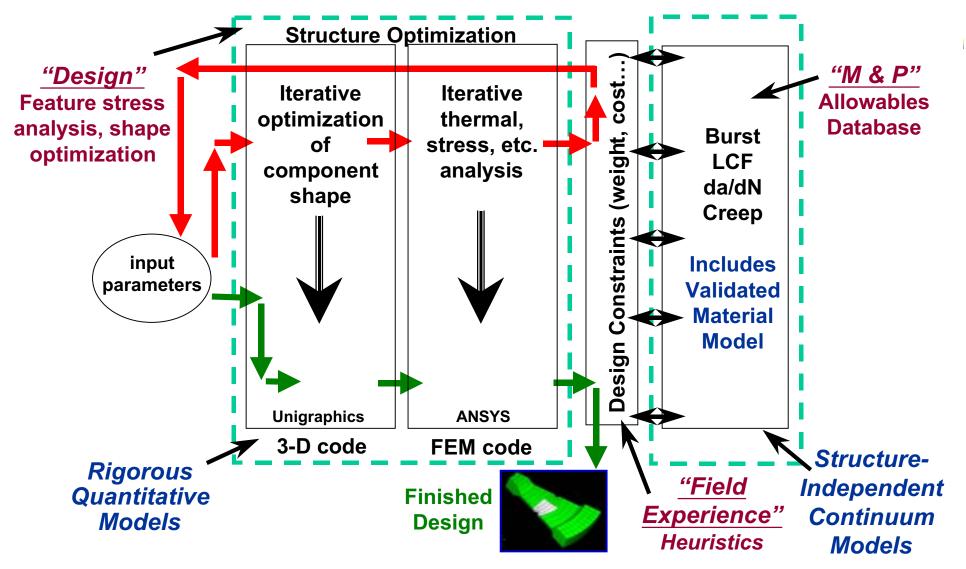
Guiding Constraints



- A key deliverable must be a validated representation of the material and process: designers work with representations!
- Structural materials design demands confidence in control of timedependent properties, thus representations needed for LCF, HCF, crack growth creep, stress rupture environmental degradation, stress corrosion friction, wear, and fretting
- 'New material' demands rapid, validated representations—but how?
- Need ubiquitous tools for optimization: a representation framework efficient validation

'Accelerated Insertion' Rather Than 'Materials by Design'

Modeling in the Component Design Process

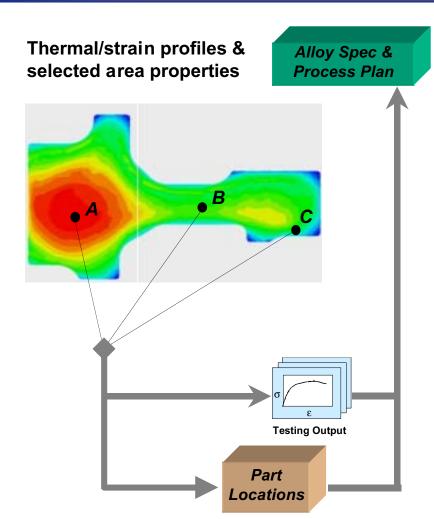


"Field Experience" corrects for i) microstructure variation, ii) inaccurate analysis, & iii) incomplete understanding of service environment



The Case of Ni-Alloy Engine Disks

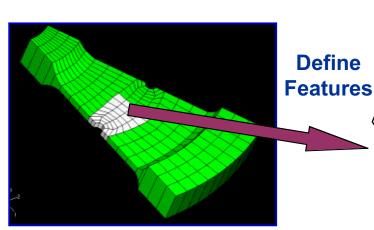




- Continuum codes (i.e., DEFORM) for thermal history and microstructure correlation over disk cross-section
- Cross-section may be "zoned" into a few regions (dual heat treat); centimeter-scale homogenization
- Empirical yield-strength models, & flow-curve 'templates,' used to assign constitutive response
- Variation of structure averaged out; local microstructure - defect interactions not represented
- Data-intensive and time-costly process for yield model and 'constitutive template' validation

Challenges to represent time-dependent failure; to introduce "new material"

The "Plasticity Engine" for Properties

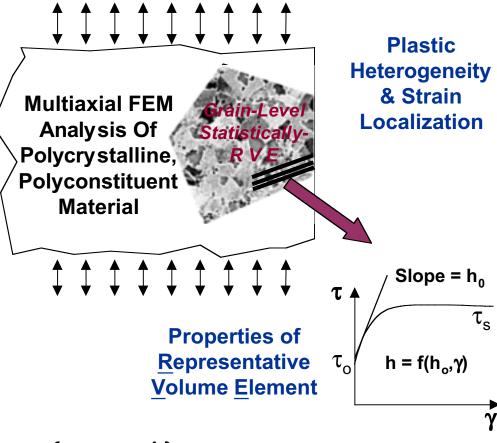


Design Concept

Focus on Work-Hardening Parameterization *With µ/s Effects*

$$\dot{\tau} = \left\{ h - \left(\frac{\tau - \tau_o}{\tau_s - \tau_o} \right) h \right\} \left(\dot{\gamma} \dot{\gamma}_o \right)^m$$

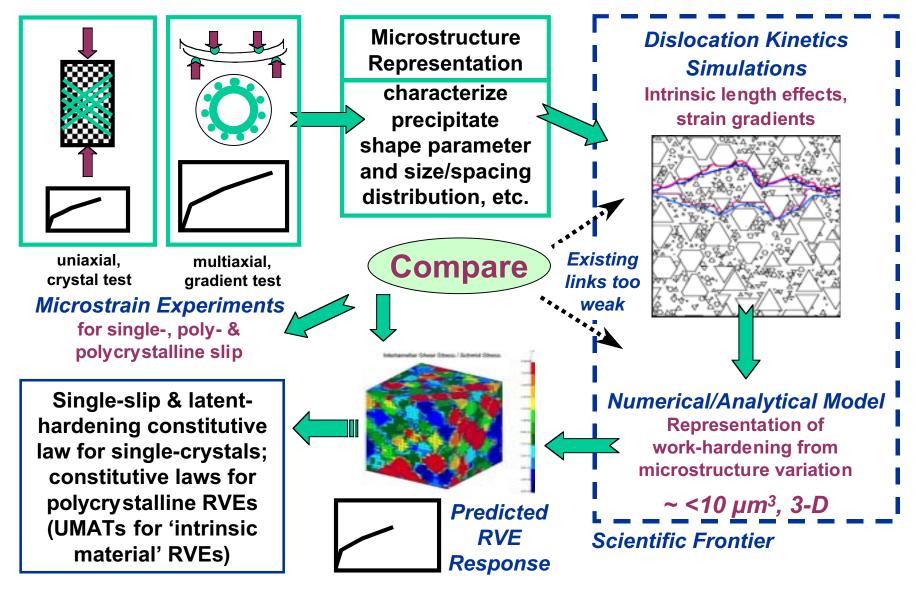
Conventional Finite Element Model Continuum Crystal Plasticity Constrained Crystal Plasticity Strain-Gradient Crystal Plasticity must define $\tau(\rho)$, $\rho(\gamma)$ to find k_o



$$\begin{split} &\{\tau_{o},\,\tau_{s},\,m,\,h\} \\ &\{\tau_{o,i},\,\tau_{s,i},\,m,\,h_{ij}\} \\ &\{\tau_{o,i}(k_{hp}),\,\tau_{s,i},\,m,\,h_{ij}\} \\ &\{\tau_{o,i},\,\tau_{s,i},\,m,\,h_{ij}\} + k_{o} \end{split}$$

increasing granularity & compute complexity

Build from Uni- / Multiaxial Slip & Work Hardening

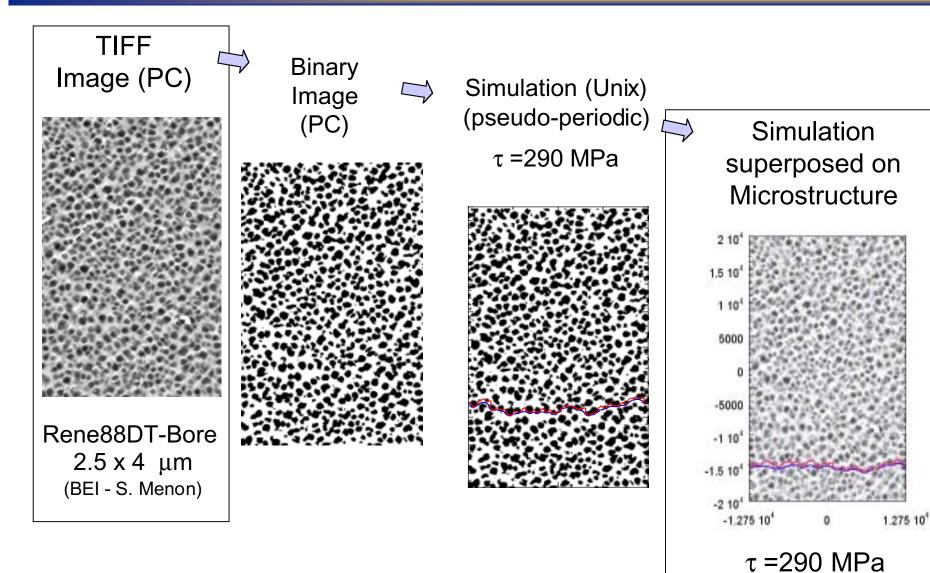


Direct links: computationally challenging & underdeveloped



Real Microstructure in Simulations





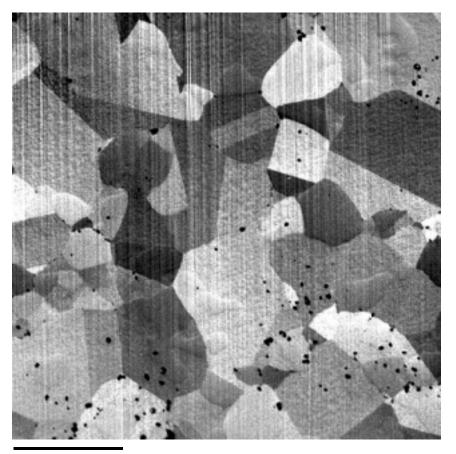
0.5



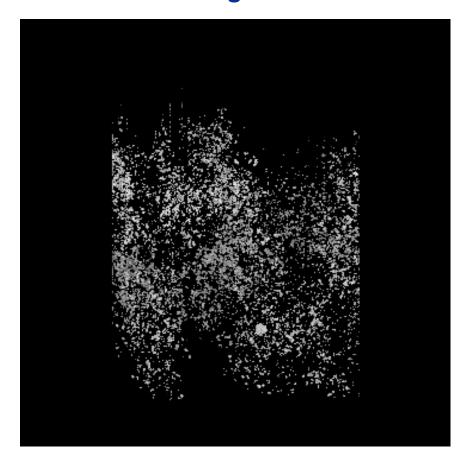
3-D Quantitative Microscopy



Serial Sectioning & Imaging



3-D Rendering of Structure



5 µm

IN-100 Ni-base Superalloy
Grain Structure and Carbide/Boride Distribution

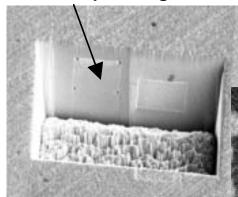


3-D Quantitative Microscopy

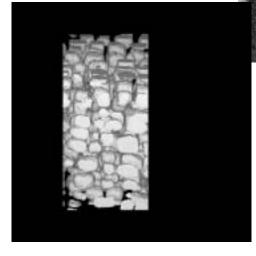


Serial Sectioning & Imaging

14 x 14 µm Image Area



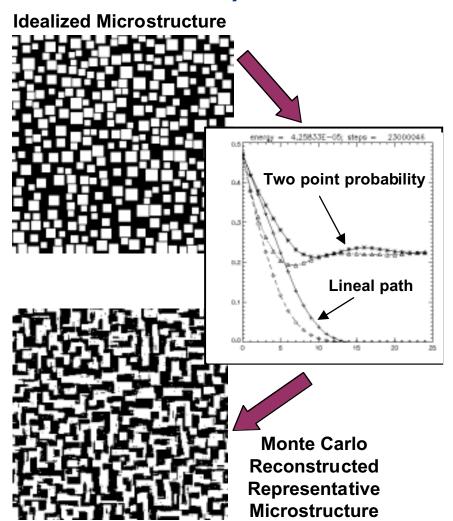
Ni-Cr-Al Superalloy



Aligned stack (~20 nm spacing)

Rendered 3-D volume (~3 µm thick)

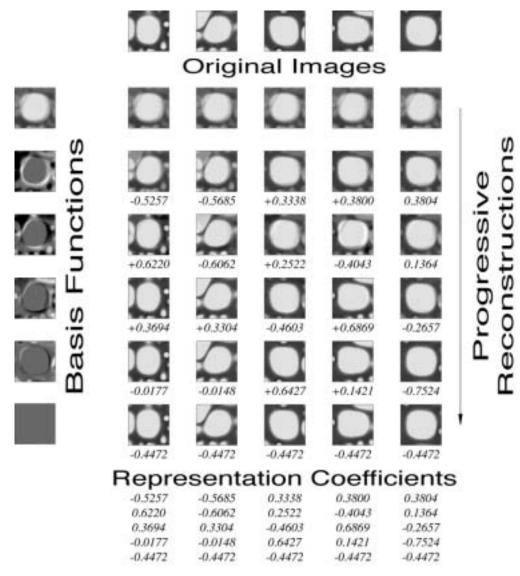
Mathematical Representation





Principal Component Analysis of Microstructure



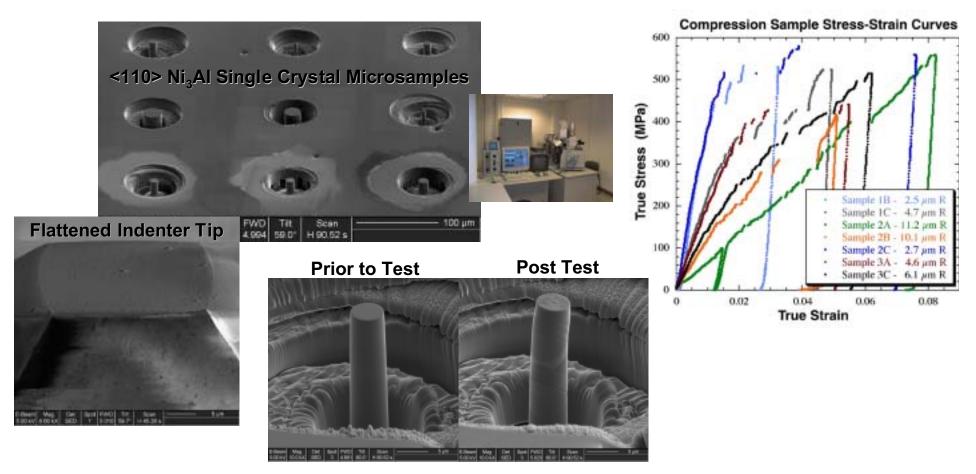




Mechanical Testing of Ultra-Small Samples for Crystal Properties



- Focus efforts on linking simulation to design
- Small-scale properties measurement for constitutive representations
- Theory for broad understanding of deformation at small scales

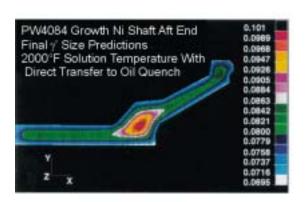


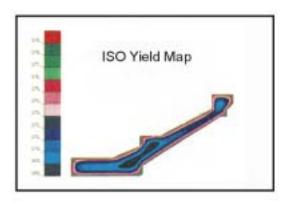


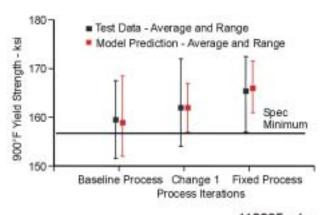
Even Simple Models Have a Big Impact



- Integrated structure-property-process models successfully applied as point solutions
 - statistically fit data to mechanistic-based property model
 - focused experiments to model microstructural evolution
 - accurate estimate of mean behavior







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P & W ----

Shaft design:

- 1/4 development time
- 80% reduction in cost

Experience shows concept is sound, projected payoffs reasonable



Eventually Must Address Full Breadth of Component Requirements

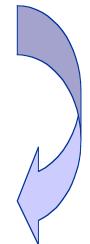


Requirements for Turbine Engine Disks:

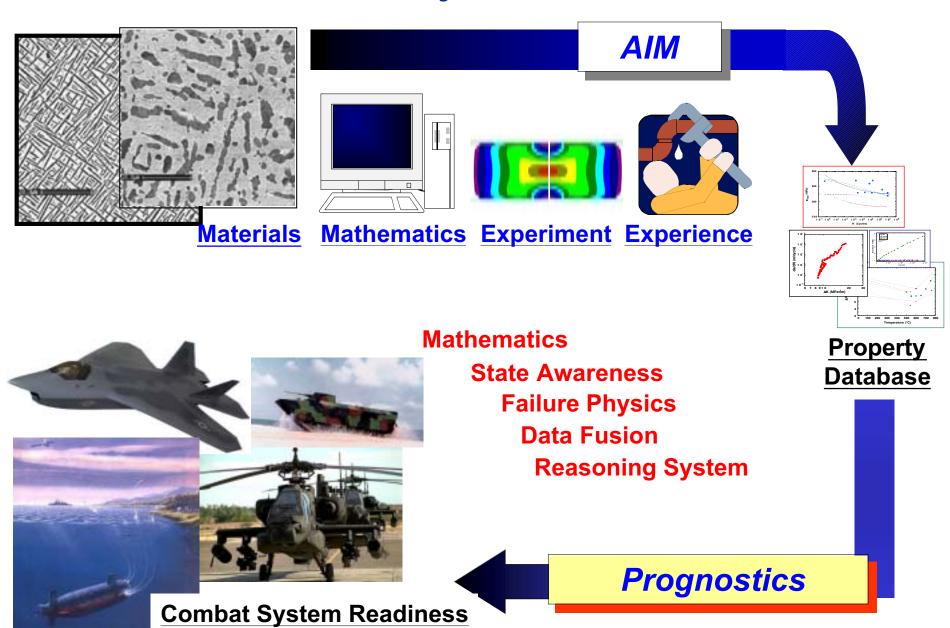
- **Ultimate Tensile Strength**
- 0.2 % Yield Strength
- Tensile Ductilities
- Notch Strength
- **Burst Margin** DARPA - AIM
- Creep
- Rupture
- Rupture Ductilities
- **Continuous Cycling LCF**
- **Hold Time LCF**
- **Continuous Cycling Crack Growth**
- **Hold Time Crack Growth**
- **Superplasticity**
- Flow Stresses
- **Abnormal Grain Growth Resistance**
- **Gamma Prime Solvus**
- Carbide(s) Solvus
- **Density**

1-Jan-04

- TIP
- **Structural Stability**
- **Exposed Behavior**
- **Defect Sensitivity**
- **Defect Content** The Issues That
- Grain Size
- Often Determine Gamma Prime Size Success or
- **Failure Segregation /Effects**
- Inspectibility
- Quench Crack Resistance
- Multi-source Capability
- Low Costs--Elemental and **Processing**
- Weldability
- Machinability
- **Machined Surface Behavior**
- Residual Stresses
- Cost Reduction Potential
- Size/Volume Scaling Effects



Materials & System Readiness





Summary



- The time for structural materials development and use must be shortened (time focus, not cost focus)
- Industrial M & P community demanding a quantum-leap in relevant engineering simulation capability
- <u>Accelerated Insertion of Materials</u> is the long-term, strategicallyrelevant, computational materials science & engineering vision
- Materials Science & Engineering community must produce integrated predictive tools
- Accelerated insertion demands integration of engineering design with M & P to achieve true systems engineering of materials technologies